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# IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

Appl. No.

10/708,110

Filed

February 10, 2004

Atty. Docket No.

02-0930A

For

Commercial Aircraft On-Board Inerting System

Date

March 3, 2006

CERTIFICATE OF FACSIMILE TRANSMISSION

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Virginia 22313-1450 on the date set forth below.

Commissioner for Patents

P.O. Box 1450

Alexandria, VA 22313-1450

Date

David Kaplan

#### SUBMISSION OF POWER OF ATTORNEY

Sir:

Please accept the following power of attorney form, and statement under 37 CFR 3.73(b), in the above-referenced patent application. Applicants hereby request that all future correspondence be directed to Customer Number 44702, Ostrager Chong Flaherty & Broitman, P.C., 250 Park Avenue, Suite 825, New York, New York 10177-0899.

Respectfully submitted,

March 3, 2006

Date

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Reg. No. 38,006

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New York, New York 10177-0899

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### MAR 0 6 2006

PTC/SB/88 (04-05)

Approved for use through 11/30/2005. ONB 0641-0035

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#### POWER OF ATTORNEY TO PROSECUTE APPLICATIONS BEFORE THE USPTO I hereby revoke all previous powers of attorney given in the application identified in the attached statement under I hereby appoint: Practitioners associated with the Customer Number: 44702 Practitioner(s) named below (if more than ten patent practitioners are to be named, then a customer number must be used): Name Registration degistration Number Number Gle<u>nn F. Ostrager</u> <u> 29,963</u> Andres Madrid 40.710 Dennis M. Flaherty <u>31,159</u> Lisa N. Benado 39,905 Joshua S. Broitman 38,006 Terje Gudmestad 32,232 Leighton K. Chong <u>Eric Satermo</u> <u>27,621</u> <u>40.159</u> Manette Dennis 30,623 John R. Rafter 28,533 as automory(s) or agent(s) to represent the undersigned before the United States Patent and Trademark Office (USPTO) in connection with any and all patent applications assigned only to the undersigned according to the USPTO assignment records or assignment documents anached to this form in accordance with 97 CFR 3.73(b). Please change the correspondence address for the application identified in the attached statement under 37 CFR 3.73(b) to: X. 44702 The address associated with Customer Mumber: OR Firm or <u>Ostrager Chong Flaherty & Broitman PC</u> Individual Name Address 250 Park Avenue, Suite 825 City New York 10177-0899 Country USA Telephone (212) 681-0600 gostrager@ocfblaw.com Assignee Name and Address: The Boeing Company 100 N. Riverside Plaza Chicago, IL 60606 A copy of this form, together with a statement under 37 CFR 3.73(b) (Form PTO/BB/95 or equivalent) is required to be filed in each application in which this form is used. The statement under 37 CFR 3.73(b) may be completed by one of the practitioners appointed in this form if the appointed practitioner is authorized to act on behalf of the assignce, and must identify the application in which this Power of Attorney is to be filed. SIGNATURE of Analyses of Record The judy idual whose signifyer and side is supplied adow is authorized to act up behalf of the assigner. Signature December 22. 2005 lerje Godmestad Telephone (949) 790-1374 Counsel, The Boeing Company

This collection of information is required by 37 CFR 1.31, 1.32 and 1.33. The information is required by obtain or retain a bossest by the public which is to file (and by the USPTO to process) an application. Confidentistip is governed by 35 U.S.C. 122 and 37 CFR 1.11 and 1.14. This collection is settlement in takes 3 minutes to complete, including exchange, preparing specific from an abundance of the complete settlement of the servour of time you require to complete this form another suppositions for restaining this burden, smalled be sent to this Chief Information Concern. U.S. Potent and Total complete. U.S. Occurrence of Commerce, P.O. Box 1450, Alexandria, VA 22313-1450. DO NOT SEND FEES OR COMPLETED FORMS TO THER ADDRESS. SEMID TO: Constitutional for Potential by the Potential VA 22313-1450. FORMS TO THIS ADDRESS. SEND TO: Commissioner for Patents, P.O. Box 1450, Alexandria, VA 22313-1450.

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STATEMENT UNDER 37 CFR 8.73(b)
Applicant/Patent Owner. The Boeing Company
Application No / Patent No.: See attached Filed/Issue Date: See attached
Entitled:
The Boeing Company a <u>corporation</u> (Name of Assignes) (Type of Assignes, e.g., corporation, purposed to, university, government agency, etc.)
states that it is; 1. X the assignee of the entire right, title, and interest, or
an assignee of less than the entire right, title and interest     (The extent (by percentage) of its ownership interest is%)
in the patent application/patent identified above by virtue of either;
A X An assignment from the inventor(s) of the patent application/patent identified above. The assignment was recorded in the United States Patent and Trademark Office at Reel Frame, or for which a copy thereof is attached.
OR  B. A chain of title from the inventor(s), of the patent application/patent identified above, to the current assignee as follows:
1. From:
The document was recorded in the United States Patent and Trademark Office at  Reel Frame or for which a copy thereof is attached.
2. From:To:To:To:
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3, From:To:To:
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Additional documents in the chain of title are tisted on a supplemental sheet.
As required by 37 CFR 3.73(b)(1)(i), the documentary evidence of the chain of title from the original owner to the assignee was, or concurrently is being, submitted for recordation pursuant to 37 CFR 3.11.
(NOTE: A separate copy (¿e., a true copy of the original assignment document(s)) must be submitted to Assignment Division in accordance with 37 CFR Part 3, to record the assignment in the records of the USPTO. See MPEP 302,08)
The understand whose titliffs supplied better the subsection behalf of the assignee.
Signature Date
Terje Gudmestad (949) 790-1374
Printed or Typed Name Telephone Number
Counsel, The Boeing Company
Title

Tris collection of information is required by 37 CFR 3.73(h). The Information is required to obtain or retain a barrels by the public which is to file (and by the USPYO to process) an application. Confidentiality is governed by 35 U.S.C. 122 and 37 CFR 1.11 and 1.14. This collection is estimated to take 12 minutes to complete, including gathering, preparing, and submitting the completed application town to the USPYO. Time will vary depending upon the impleted complete this form another suggestions for reducing this burden, should be sets to the Chief information Officer, U.S. Palent and Tradewart Office, U.S. Department of Commerce, P.O. Box 1460, Alexandria, VA 22313-1450. DO NOT SEND FEES OR COMPLETED FORMS TO THIS ADDRESS. SEND TO: Committee for Partents, P.O. Box 1469, Alexandria, VA 22313-1450.

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	WIDE-BANDGAP, LATTICE-MISMATCHED WINDOW LAYER FOR A SOLAR ENERGY	09/976,508	12-Oct-01		
		1001010100			
_J		1	12 000	1012271	0098
	CONVERSION DEVICE	į.	İ	•	1
Α	WIDE-BANDGAP, LATTICE-MISMATCHED	10/356,028	31-Jan-03	1014250	0577
į	WINDOW LAYER FOR A SOLAR ENERGY	12,230,020	) July	014203	0077
	CONVERSION DEVICE	l	i	į	1
1	ANTENNA FEEDFORWARD INTERFERENCE	09/853 475	11-May-01	011809	0297
<u>i</u>	CANCELLATION SYSTEM	İ	1	1011005	025
T	SEMICONDUCTOR CIRCUITS AND DEVICES	09/850 773	DR-May 01	011792	0263
	ON GERMANIUM SUBSTRATES		100 11/23		0200
C	Liquid Hydrogen Fueled Aircraft with High Wing	29/189,740	10-Sep-03	016149	0392
1	Method and System for Reducing Stress				0545
Ì	Concentrations in Lap Joints	1		10.000	100.0
1	Method and System for Utilizing Low Pressure	10/404.742	01-Apr-03	013938	0241
•	for Perforating and Consolidating an Uncured		1		J
<u> </u>	Laminate Sheet in One Cycle of Operation	1	<u> </u>	j	1
Α	Low Chamfer Angled Torque Tube End Fitting	10/710.645	27-Jul-04	014899	0101
<u>!</u>	With Elongated Overflow Groove		}		
. j <u> </u>	Simulation System And Method	09/865,293	25-May-01	011860	0356
	Dual-Band Multiple Beam Antenna System For				0533
		1			1
Α		11/259,913	27-Oct-05	012557	0533
<u> </u>		}	1		
ļ,		10/137,974	03-May-02	012869	0731
ļ	Aircraft Surface Ice Inhibitor	10/161,238		013209	0635
Ļ	A Method for Detecting Foreign Object Debris			012181	0775
:	Operating Point Independent Digital Automatic	10/389,034			0735
ļ		<u> </u>	1		1
	Redundant Power Distribution System		09-Jul-03	014267	0982
į		10/349,294	22-Jan-03	013693	0930
ļ <b>-</b>			<u>L</u>		
}	Method and System Having a Flowable	10/404,993	01-Apr-03	013938	0234
ļ	Pressure Pad for Consolidating an Uncured	1			l
<del> </del> -					
!	Thermographic System and Method for	10/274,273	18-Oct-02	014219	0150
}	Detecting Imperections within a Bond				<u> </u>
_	Operational Ground Support System		17-May-04	015160	0505
	Operational Ground Support System				0354
<b>-</b> .	Count Support Contains for an Operational	11/163,405	18-Oct-05	016655	0986
	Ground Support System				
Ì		10/397,003	25-Mar-03	013918	0156
		274445			<b></b>
		10/142,461	10-May-02	012899	0867
				<del></del>	<u> </u>
	Volume Propellent Tente	10/327,317	20-Dec-02	013618	0959
	Rechampable Composite Dhe Application	401076.665	400		
Α	Rechargeable Composite the Assistance				0926
					0926
		10/33/,530	01-19U-03	U13644	0043
		10/226 964	00 0 00	040076	0000
	Applications	10/230,301	vo-sep-uz	V132/6	0573
		CANCELLATION SYSTEM  SEMICONDUCTOR CIRCUITS AND DEVICES ON GERMANIUM SUBSTRATES C Liquid Hydrogen Fueled Aircraft with High Wing Method and System for Reducing Stress Concentrations in Lap Joints Method and System for Utilizing Low Pressure for Perforating and Consolidating an Uncured Laminate Sheet in One Cycle of Operation A Low Chamfer Angled Torque Tube End Fitting With Elongated Overflow Groove Simulation System And Method Dual-Band Multiple Beam Antenna System For Communication Satellites Electronic Network Filter for Classified Aircraft Surface Ice Inhibitor A Method for Detecting Foreign Object Debris Operating Point Independent Digital Automatic Level Control Redundant Power Distribution System Closed-Loop Pointing System with Spot Beams and Wide-Area Beams Method and System Having a Flowable Pressure Pad for Consolidating an Uncured Laminate Sheet In a Cure Process Thermographic System and Method for Detecting Imperfections within a Bond Operational Ground Support System E Cerry-On Luggage System for an Operational Ground Support System Low-Penetration-Force Pinnat for Perforating an Uncured Laminate Sheet Multi-Dimensional Fractional Number of Bits Modulation Scheme Increased Propellant Performance From Equal Volume Propettant Tanks Rechargeable Composite Ply Applicator Dual Transmission Emergency Communication System Improved Honeycomb Cores For Aerospace	CANCELLATION SYSTEM  SEMICONDUCTOR CIRCUITS AND DEVICES ON GERMANIUM SUBSTRATES  C. Liquid Hydrogen Fueled Aircraft with High Wing Method and System for Reducing Stress Concentrations in Lap Joints Method and System for Utilizing Low Pressure for Perforating and Consolidating an Uncured Laminate Sheet in One Cycle of Operation  A. Low Chamfer Angled Torque Tube End Fitting With Elongated Overflow Groove Simulation System And Method Oual-Band Multiple Beam Antenna System For Communication Satellites  A. Dual-Band Multiple Beam Antenna System For Communication Satellites  Dual-Band Multiple Beam Antenna System For Communication Satellites Electronic Network Filter for Classified Aircraft Surface Ice Inhibitor Aircraft Surface Ice Inhibitor Operating Point Independent Digital Automatic Level Control Redundant Power Distribution System Operating Point Independent Digital Automatic Level Control Redundant Power Distribution System Object Debris Object Debris Operating Foreign Object Debris Operating Point Independent Digital Automatic Level Control Redundant Power Distribution System 10/349,294 and Wide-Area Beams Method and System Having a Flowable Pressure Pad for Consolidating an Uncured Laminate Sheet In a Cure Process Thermographic System and Method for Detecting Imperfections within a Bond Operational Ground Support System Operational Ground Support System Low-Penetration-Force Pinmat for Perforating an Uncured Laminate Sheet Multi-Dimensional Fractional Number of Bits Modulation Scheme Increased Propellant Performance From Equal Volume Propellant Tanks Rechargeable Composite Pty Applicator 10/237,317 Volume Propellant Tanks Rechargeable Composite Pty Applicator 10/236,381	CANCELLATION SYSTEM SEMICONDUCTOR CIRCUITS AND DEVICES 09/850,773 08-May-01 ON GERMANIUM SUBSTRATES C Liquid Hydrogen Fueled Aircraft with High Wing 29/189,740 10-Sep-03 Method and System for Reducing Stress 10/905,484 06-Jan-05 Concentrations in Lap Joints Method and System for Utilizing Low Pressure for Perforating and Consolidating an Uncured Laminate Sheet in One Cycle of Operation A Low Chamfer Angled Torque Tube End Fitting With Elongated Overflow Groove Simulation System And Method 09/865,293 25-May-01 Dual-Band Multiple Beam Antenna System For 10/060,822 30-Jan-02 Communication Satellities Electronic Network Filter for Classified 10/137,974 03-May-02 Aircraft Surface Ice Inhibitor 10/161,298 31-May-02 Aircraft Surface Ice Inhibitor 10/1615,705 09-Jul-03 Closed-Loop Pointing System with Spot Beams 10/349,294 14-Mar-03 Level Control Redundant Power Distribution System 10/615,705 09-Jul-03 Closed-Loop Pointing System with Spot Beams 10/349,294 22-Jan-03 and Wide-Area Beams Method and System Having a Flowable Pressure Pad for Consolidating an Uncured Laminate Sheet in a Cure Process Thermographic System and Method for Detecting Imperfections within a Bond Operational Ground Support System 10/711,610 28-Sep-04 Cerry-On Luggage System for an Operational Ground Support System Low-Penetration-Force Pinmat for Perforating 10/397,003 25-Mar-03 an Uncured Laminate Sheet Multi-Dimensional Fractional Number of Bits Modulation Scheme Increased Propellant Tanks Rechargeable Composite Ply Applicator 10/272,085 16-Oct-02 A Rechargeable Composite Ply Applicator 10/273,537 20-Dec-02 Volume Propellant Tanks Rechargeable Composite Ply Applicator 10/236,381 08-Sep-02 Increased Propellant Tanks Rechargeable Composite Ply Applicator 10/236,381 08-Sep-02 Increased Propellant Sep-04 Communication System Improved Honeycomb Cores For Aerospace 10/236,381 08-Sep-02 Increased Propellant Sep-04 Communication 10/236,381 08-Sep-02 Increased Propellant Sep-04 Communication 10/236,381 08-Sep-02 Increased Propellant Sep-04 Communication 1	CANCELLATION SYSTEM  SEMICONDUCTOR CIRCUITS AND DEVICES 09/850,773 08-May-01 011792 ON GERMANIUM SUBSTRATES  C Liquid Hydrogen Fueled Aircraft with High Wing 29/189,740 10-Sep-03 016149 Method and System for Reducing Stress Concentrations in Lap Jolints Method and System for Utilizing Low Pressure for Perforating and Consolidating an Uncured Laminate Sheet in One Cycle of Operation A Low Chamfer Angled Torque Tube End Fitting With Elongated Overflow Groove Simulation System And Method 09/865,293 25-May-01 011860 Dual-Band Multiple Beam Antenna System For Communication Satellities  Dual-Band Multiple Beam Antenna System For Communication Satellities  Electronic Network Filter for Classified 10/137,974 03-May-02 012557 Communication Settlities  Electronic Network Filter for Classified 10/137,974 03-May-02 012869 Aircraft Surface Ics Inhibitor 10/161,238 31-May-02 013208 A Method for Detecting Foreign Object Debris Operating Point Independent Digital Automatic Level Control  Redundant Power Distribution System 10/615,705 09-Jul-03 014267 Closed-Loop Pointing System with Spot Beams and Wide-Area Beams Method and System Harving a Flowable Pressure Pad for Consolidating an Uncured Laminate Sheet In a Cure Process Themographic System and Method for Detecting Imperfections within a Bond Operational Ground Support System 10/647,739 17-May-04 015169 Cearry-On Luggage System for an Operational Ground Support System 10/0397,003 25-Mar-03 013918 United Dimensional Fractional Number of Bits 10/142,461 10-May-02 012899 Motuation Scheme Increased Propellant Performance From Equal Volume Propellant Tanks Rechargeable Composite Pty Applicator 10/236,361 08-Sep-02 013376 Improved Honeycomb Cores For Aerospace 10/236,361 08-Sep-02 013376

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02-0667		Communication System for Tracking Assets	10/310,457	05-Dec-0		0810
02-0714		Robust Palladium Based Hydrogen Sensor	10/382,187	05-Mar-0:		0309
02-0718	1	Optical Differential Quadrature Phase-Shift	10/281,676			0036
	;	Keyed Decoder	101201,010	20-000-0	2013434	0030
02-0889	-	Constant Vertical State Maintaining Cueing	10/613,253	03 54 0	014295	
	Ì	System	10010,200	) 03-Jui-03	3014295	0258
02-0930	A	COMMERCIAL AIRCRAFT ON BOARD	407700 445	<del></del>	1	- <del>  </del>
OL DODO		INERTING SYSTEM	10/708,110	10-Feb-04	014318	0304
02-1095	+	Programmable Messages for Communication		i	l	
02-1023	İ	Programmable Messages for Communication	10/310,275	05-Dec-02	013554	0714
22 4000	<del>;</del>	System having One-Button User Interface			L	i
02-1096	<u> </u>	Communications Protocol for Mobile Device	10/310,481	05-Dec-02	013554	0606
02-1150	•	On Orbit Variable Power High Power Amplifiers	10/365,359	12-Feb-03	013764	0001
	. <b>i</b> ,	for a Satellite Communications System	Ĺ	ì	1	
02-1189	į	VARIABLE HIGH POWER AMPLIFIER WITH	10/431,903	08-May-03	014060	0978
	ì	CONSTANT OVERALL GAIN FOR A		"	1	1
	! 	SATELLITE COMMUNICATION SYSTEM	1	į.	1	į.
02-1221		Serial Port Multiplexing Protocol	10/310,751	05-Dec-02	1013553	0935
02-1231	1	METHOD FOR PREPARING ULTRA-FINE.	10/707,173	25-Nov-03		0797
	ì	SUBMICRON GRAIN TITANIUM AND	101101,110	20101-0	1014133 	lotai
	1	TITANIUM-ALLOY ARTICLES AND ARTICLES	}	i	1	1
	Į.	PREPARED THEREBY	ļ	}	i	Ì
2-1244	`` !	Fiber Matrix for a Geometric Morphing Wing	40007 000	1 20 5	24222	<del> </del>
2-1264	!·	Resonator Box to Laser Cavity Interface for	10/357,022	03-Feb-03		0097
1207	l	Chemical Laser	10/396,804	24-Mar-03	013914	0840
2-1300	<del> </del> -		<del> </del>	İ	<u> </u>	
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2-1349	<b></b> -	Foreign Object Debris	<u> </u>		<u>L</u>	L
3-0030	<u> </u>	Integrated Window Display	10/383,012	06-Mar-03		0001
J3-003Q	i	PPM RECEIVING SYSTEM AND METHOD	10/707.076	19-Nov-03	014140	0908
	ļ. —	USING TIME-INTERLEAVED INTEGRATORS				1
3-0138	<u> </u>	Capacitive Acceleration Derivative Detector	10/604,537	30-Jul-03	013834	0446
3-0192	İ	AUTONOMOUSLY ASSEMBLED SPACE	10/605,797	28-Oct-03		0717
	<u> </u>	TELESCOPE				1
	A_	Fast Access, Low Memory, Pair Catalog	10/710,177	24-Jun-04	014769	0432
3-0198		Method and Apparatus for Real-Time Star	10/709,346	29-Арг-04		0263
	i	Exclusion From A Database		,	0.7007	0200
3-0197	Α	Method and Appartus For On-Board	10/710,17B	24-Jun-04	014760	0735
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3-0208		Variable-Duct Support Assembly	10/708,864	29-Mar-04	044457	0000
3-0271		BEAMFORMING ARCHITECTURE FOR MULTI	10/705,004	26-Nov-03		0228
		BEAM PHASED ARRAY ANTENNAS	14701,211	Z0-140V-U3	U1415 <del>9</del>	0794
3-0348	<u> </u>	Aircraft Interior Configuration Detection System	40540 007			
3-0414			10/710,287	30-Jun-04		0966
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<del></del>		Aircraft Secondary Electric Load Controlling	10/604,189	30-Jun-03	013765	0377
2.0400		System	<u> </u>			Ĺ
3-0489		GPS NAVIGATION SYSTEM WITH	10/605,890	04-Nov-03	014100	0958
		INTEGRITY AND RELIABILITY MONITORING		•		
3-0520		Integrated Capacitive Bridge Integrated Flexure	10/953,726	29-Sep-04	015837	0448
i		Functions Inertial Measurement Unit	}			
3-0527		Dynamic Seat Labeling and Passenger	10/707,965	28-Jan-04	14287	0001
:		Identification System		-2 201-04		WU 1

<b>E</b> 32 V			4.6		1 62 3	The Sylvenia
03-0684	1	Integral Clamping-and-Bucking Apparatus for	10/904,978	08-Dec-04		
		Utilizing a Constant Force and Installing Rivet	1	10000	10.0124	10302
ļ ————————	<u>.</u>	Fastenets in a Sheet Metal Joint	ļ	1	1	ļ
03-0755	<del> </del>	Heavy Particle Lorentz Force Accelerator	10/709.620	18-May-04	014623	0324
03-0835		Aircraft Archway Architecture	10/688,624			0753
03-0835	A	Interior Archway for an Aircraft	29/192,055			0075
03-0835	B	Aircraft Interior Architecture	10/908,140	28-Apr-0		0075
03-0835	ic .	Modular Archway for an Aircraft	29/228,800		014628	0075
03-0885	1	Lightweight Composite Fairing Bar and Method	11/160,192	13-Jun-05		0060
	<u> </u>	for Manufacturing the Same		1	]*.0.02	10000
03-0925		Interior Seating Architecture for Aircraft	10/805,586	10-Oct-03	014040	0514
03-0963	!	MULTIPLE STAYOUT ZONES FOR GROUND	10/709,348			0363
	<b>-</b>	BASED BRIGHT OBJECT EXCLUSION	1	1	0.300.	4445
03-1090	ļ	Translucent, Flame Resistant Composite	10/707,612	24-Dec-03	014217	0512
	L	Materials			1721,	05.2
03-1104	} 	Shower System	10/708,749	23-Mar-04	014440	0233
03-1129	į	Unauthorized Access Embedded Software	10/658,159	09-Sep-03		0326
	<u>'</u>	Protection System		1 22 20p 30	0.4430	0320
03-1138		Undercut for Bushing Retention for SLS Details	10/710,144	22-Jun-04	014760	0698
03-1140	<del></del> _	SLS for Tooling Applications	10/710,163			0205
03-1308		Mandrel, Mandrel Removal and Mandrel	10/907,320			0315
		Fabrication to Support a Monolithic Nacelle	1	20-11121-00	013636	0315
		Composite Panel	1			
03-1471		Extended Accuracy Variable Capacitance	10/952,952	29-Sep-04	NIEREE	0647
<u></u>		Bridge Accelerometer	147002,002	20 Ocpus	013000	0047
03-1526		Flexible Mandrel for Highly Contoured	10/904,717	24-Nov-04	045204	0571
		Composite Stringer	1	2-1-104-04	019391	U\$/1
04-0016	À	AN INTEGRATED TRANSPORT SYSTEM AND	10/709 777	27-May-04	O44REA	0676
İ		METHOD FOR OVERHEAD STOWAGE AND	100,000,777	ZI-IWay-O-	014004	00/0
:		RETRIEVAL	į .			ļ ·
04-0054	Ā	REAL-TIME REFINEMENT METHOD OF	11/028,094	03-Jan-05	018178	0162
i		SPACECRAFT STAR TRACKER ALIGNMENT	1,020,034	vo-vair-us	010170	0162
		ESTIMATES				
04-0070	•	Enhanced Pinmat for Manufacturing High-	10/904,012	19-Oct-04	046067	0000
		Strenth Perforated Laminate Sheets	10004,012	15-00-04	U15267	0039
04-0072		Overhead Space Access Conversion Monument	10/708 810	26-Mar-04	044454	0789
		and Service Area Staircase and Stowage	14700,010		U 1445 I	0/69
04-0073		Stowable Spiral Staircase System for Overhead	10/708 855	29-Mar-04	N444E7	0168
[		Space Access	10.00,000	23-11101-01	V 14401	0100
14-0089		Determinant Assembly Features for Vehicle	10/904,802	30-Nov-04	015200	0122
		Structures	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		013333	0122
4-0092		Overhead Space Access Stowable Staircase	10/708,733	22-Mar-04	014435	0168
4-0097		MANDREL WITH DIFFERENTIAL IN	10/904,709	24-Nov-04		0450
	i	THERMAL EXPANSION TO ELIMINATE	.000	24-1404-04	013331	V450
4-0137		Method to Improve Properties of Aluminum	10/939,528	13-Sep-04	016625	0434
i		Alloys Processed by Solid State Joining		. O Cap Cap	~ \$0000	454
4-0208		Segmented Flexible Barrel Lay-up Mandrel	10/904,841	01-Dec-04	M5404	0307
4-0304		Mist Delivery System				0637
4-0384		Self-Locating Feature for a PI-Joint Assembly	10/904,800	30-Nov-04		
4-0385	i	Minimum Bond Thickness Assembly Feature	10/904,801	30-Nov-04 (		0995 0046
		Assurance			10024	UV+0
4-0567	T	Aircraft Cabin Crew Complex	10/711,386	15-Son M/	115120	0758
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04-0588		Articulated Spacecraft Seat and Stretcher	10/906,482	22-Feb-05	045604	0268
04-0589	]	Composite Shell Spacecraft Seat	10/905,483	06-Jan-06		0975
04-0590	7	Adjustable Attenuation System for a Space Re-	10/907,931	21-Apr-05		0242
<u> </u>	į	Entry Vehicle Seat	10001,501	E I TOPI TO	1013326	0242
04-0667	į.	Airport Security System	10/905,757	04-Mar-05	1016720	0856
04-0681		Protective Cover and Tool Splash for Vehicle	10/907,786	15-Apr-05		0530
ļ	1	Components	100001,100	10-241-00	0 13804	10030
04-0741		Pivot Mechanism for Quick Installation of	10/905,502	07-Jan-05	048545	0015
		Stowage Bins or Rotating Items	{	) UNAIPOS	0 13343	פוטטן
04-0747	1	Stowable Table	10/907,600	07-Apr-05	045975	0804
04-0765	7~~	Laycred, Transparent Thermoplastic for	11/102,401	08-Apr-05	015013	0082
<u> </u>	1	Flammability Resistance	, 102,401	oo spra	010003	0002
04-0791	<u> </u>	Electromagnetic Mechanical Pulse Forming of	10/905,211	21-Dec-04	015477	0601
i	į_	Fluid Joints for High-Pressure Applications	1.0000,211	21-060-04	013477	נתפתו
04-0793		Airplane Interior Systems	10/907,990	22-Apr-05	015000	0923
04-0805	-	Compensated Composite Structure	10/994,848			0742
04-0824	1	Aircraft Cart Transport and Stowage System	10/906,465			
04-0859	7	Magnetic Null Accelerometer	10/905,007	09-Dec-04		0473
04-0893	<u> </u>	In-Process Vision Detection of Flaws and FOD	10/904,719			0879
į	i	By Back Field Illumination	10/304,7 13	24-1404-04	ופכבוטן	0395
04-0914	Ţ	Aircraft Sink with Integrated Waste Disposal	10/907,625	08-Apr-05	045077	
	į	Function	10/30/,023	00-Apr-03	119611	0782
04-0977	T	Extended Accuracy Flexured Plate Dual	10/907,751	14-Apr-05	040070	-
	Ì	Capacitance Accelerometer	10/20/,/51	14-7401-05	016279	0012
04-0993	1	Design Methodology to Maximize the	10/907,973	22 454 05	74500G	2500
		Application of Direct Manufactured Aerospace	100901,813	22-Apr-05	010933	0523
04-0993	A	Flow Optimized Stiffener for Improving Rigidity	11/162,261	02-Sep-05	040400	00.57
	İ	of Ducting	11102,201	02-36p-03	010490	0847
04-1054	· · · · ·	Electromagnetic Mechanical Pulse Forming of	11/028,093	03-Jan-05	045470	0744
	Í	Fluid Joints for Low-Pressure Applications	17020,000	OS-Jair-OS	סווסוט	0741
04-1137	i	Jet Airplane Configuration	29/220,256	28-Dec-04	040040	000=
04-1137	Α	Jet Airplane Configuration	29/220,254			0260
04-1137	В	Jet Airplane Configuration	29/220,255			0953
04-1240	<del>                                     </del>	Method and Apparatus for Optically Detecting	11/164,414	28-Dec-04 22-Nov-05		0268
		and Identifying a Threat	1	22-1404-05	O 1000B	0671
04-1256		Multi-Ring System for Fuselage Formation	10/907,729	13-Apr-05	045000	2018
04-1263	1	Integrally Damped Composite Aircraft Floor	11/163,957	04-Nov-05		0016 0779
		Panels	1	01101-05	010732	0779
05-0020		Integrated Wiring for Composite Structures	11/163,001	30-Sep-05	046806	0244
05-0084		Aircraft Stowage Bln	11/163,801	31-Oct-05		0199
05-0164		Multiple Attendant Galley	11/160,958	18-Jul-05		0577
05-0263	·		11/161,735			0090
	!	Transportation, and Storage of Large Shell		io-rug-ou	V 10-10-5	OUSU
	L	Structures	İ	1		
05-0288		Stringer Holding Device	11/162,257	02-Sep-05	018400	0528
05-0300		Ceiling Illumination for Aircraft Interiors	11/164,267	18-Nov-05		0183
5-0302		Collapsible Guide for Non-Automated Area	11/161,769	16-Aug-05		0593
		Inspections		. V . My UO		~33
05-0355		Antenna Vibration Isolation Mounting System	11/164,309	17-Nov-05	016795	0416
5-0360		Renewable Superhydrophobic Coating	11/160,600	30-Jun-05		0284
5-0377		Flow Path Splitter Duct	11/163,137	06-Oct-05 (		0041
5-0402	_	Rotor/Wing Dual Mode Hub Fairing System	11/162,924	28-Sep-05 (		0959

05-0410	Dehumidifying Radome Vent	11/164,225	(A) (A) (A)		
05-0466	Environmentally Stable Hybrid Fabric System for Exterior Protection of an Aircraft	11/163.614		1	0030 0681
05-0493 05-0541	Space Depot For Spacecraft Resupply Anti-Personnel Airborne Radar Application	11/162,333			0797
05-0624	An Uptoaded Lift Offset Rotor System For A Helicopler	11/162,474	12-Sep-05 18-Oct-05		0855 0683
05-0723	Method to Control Thickness in Composite Parts Cured on Closed Angle Tool	11/164,103	10-Nov-05	016762	0663